

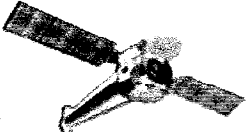


Operations

CHANDRA

Chandra Operations

**Robert Cameron
Chandra X-ray Observatory Center**



Operations

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Radiation-related Operations:

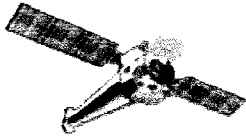
The objective: protecting the Science Instruments, while maximizing the science observing time on Chandra.

The tasks:

- Implement autonomous radiation protection on Chandra
- Implement a radiation monitoring and alert system for the Science Operations Team and Flight Operations Team
- Improve radiation prediction models for mission planning

The groups involved:

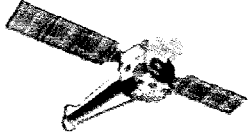
- Science Operations Team. Leader: Dan Schwartz
- Radiation Working Group. Leader: Steve O'Dell
- Flight Operations Team, Mission Planning. Leader: Dan Shropshire



SI Safing Enhancements

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-
- Implemented in November 1999.
 - Provided an autonomous safing capability for protection of Science Instruments. This does not replace scheduling of Science Instrument protection activities through perigee passages, but provides backup protection in case of schedule failures (e.g. if the mission schedule is incorrectly built or stops running).
 - Extended the spacecraft safing capabilities to include safing of the science instruments in the event of spacecraft failures interrupting the science mission.
 - Provided better use of EPHIN energy ranges, for detecting low energy protons, and adjusted the particle count rate thresholds to prevent false triggers outside perigee while providing reliable detection of potentially damaging proton events.
 - Accompanied by improved Flight Operations Team monitoring of science instrument safety, and improved ground-based science instrument safing procedures (including an OCC script-based version of the autonomous procedure).
 - Accompanied by improved review of mission command loads to check correct science instrument safing command sequences and times for radiation belt passages.



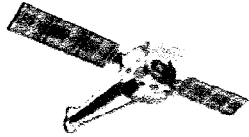
SI Safing Enhancements

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A universal instrument safing response applies to all SI safing triggers. For autonomous safe operation, the mission command loads must be stopped. The same safing thresholds are applied during ACIS and HRC observations. There is no autonomous recovery from instrument safing.

SI safing triggering events:

- Observatory Safe Mode
- Normal Sun Mode transition
- Bright Star Hold
- RADMON high radiation detection
- EPHIN failure
- On-board computer Standby Timeout



SI safing actions, implemented in Stored Command Sequence 107:

1. Disable RADMON Process
2. Terminate Mission SCSs
3. Disable Momentum Unloading; disable VDE MUPS valve power
4. Disable Spacecraft Momentum Monitor Safing
5. Disable RW Momentum Bias
6. Retract OTGs
7. Move SIM Translation Table to midpoint
8. Safe ACIS: stop science; power down VBs and FEPs; dump system configuration
9. Safe HRC: stop I&S HV ramp up and HRC dither control; I&S HV rampdown; A&B shield HV turnoff; retract focus shutters; open HRC door; mostly close HRC door
10. Move SIM Translation Table to HRC-S Position
11. Enable Momentum Unloading
12. Enable Spacecraft Momentum Monitor Safing
13. Enable RW Momentum Bias
14. Disable SCS 107



SI Safing Enhancements

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SI Safing Events

17 February 2000: Chandra Safe Mode; ground command error.

5 June 2000: SCS 107 executed by ground command, because of incorrect command timing in mission load.

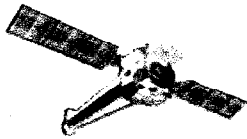
8 June 2000: Autonomous; high radiation.

13 July 2000: SCS 107 executed by ground command; high radiation.

13 September 2000: Autonomous; high radiation.

8 November 2000: Autonomous; high radiation.

26 November 2000: Autonomous; high radiation.

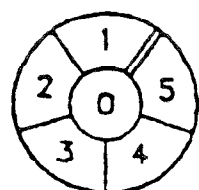


EPHIN

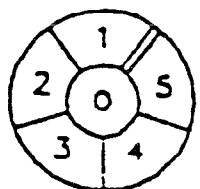
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Electron Proton Helium Instrument (EPHIN)

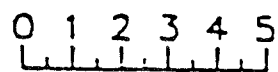
- Designed for the Solar and Heliospheric Observatory (SOHO), by University of Kiel. Principal Investigator: Reinhold Mueller-Mellin.
- Stack of 5 Si detectors + plastic anti-co shield + sixth Si detector (for penetrating particles).
- 83 degree field of view.
- 5.1 cm²-sr geometric factor.
- A and B detectors have switchable geometry, for high count rates. Geometric factor changes by 24x, which allows flux measurements up to 200000 cts/cm²-s-sr without significant dead time losses.
- Telemetry update period: 65.6 seconds.
- Chandra uses electron and proton rates for on-board radiation monitor.



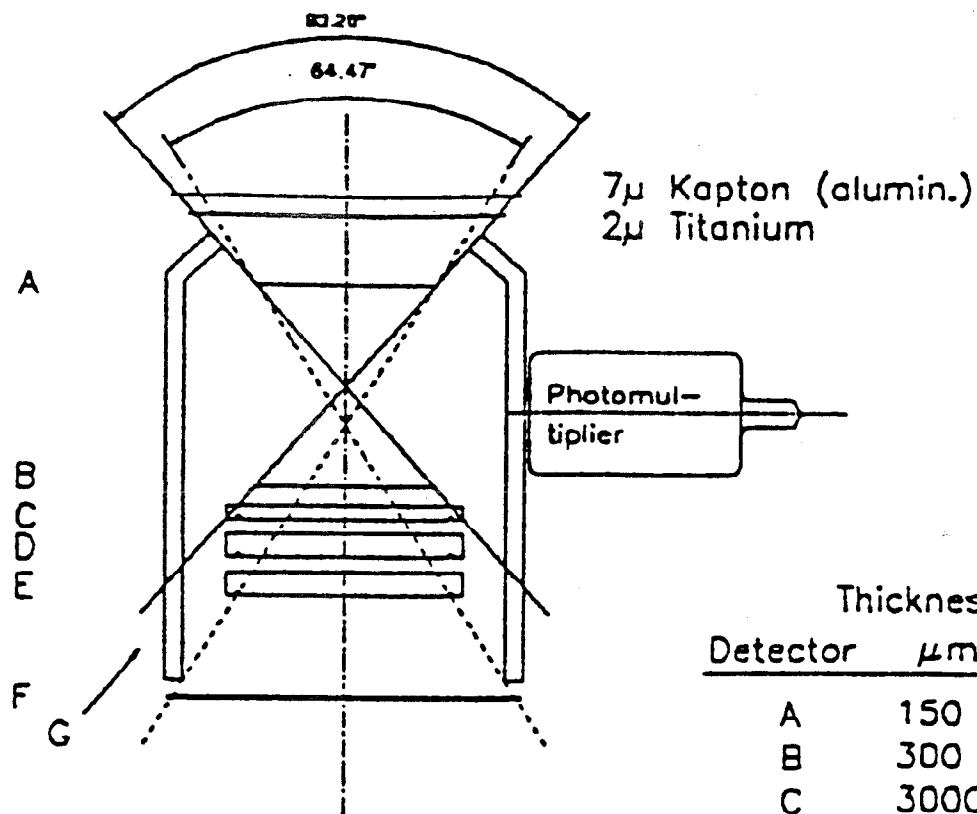
A



B



scale (cm)



EPHIN

Detector	Thickness μm	Active Area cm^2	Type
A	150	11.3	ion-implanted
B	300	11.3	ion-implanted
C	3000	15.2	lithium-drifted
D	5000	15.2	lithium-drifted
E	5000	15.2	lithium-drifted
F	700	50.0	ion-implanted
G	4000		plastic scintill.



EPHIN

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EPHIN Channels

<u>Type</u>	<u>Name</u>	<u>Energy Range</u> (MeV)	<u>Energy Width</u> (MeV)
Electron	E150	0.25 – 0.70	0.45
	E300	0.67 – 3.00	2.3
	E1300	2.64 – 6.18	3.6
	E3000	4.80 – 10.4	5.6
Proton	P4	5.0 – 8.3	3.3
	P8	8.3 – 25.0	16.7
	P25	25.0 – 41.0	16.0
	P41	41.0 – 53.0	12.0
Helium	H4	5.0 – 8.3	3.3
	H8	8.3 – 25.0	16.7
	H25	25.0 – 41.0	16.0
	H41	41.0 – 53.0	12.0
Integral	INT	e>8.7, p>53, He>53	



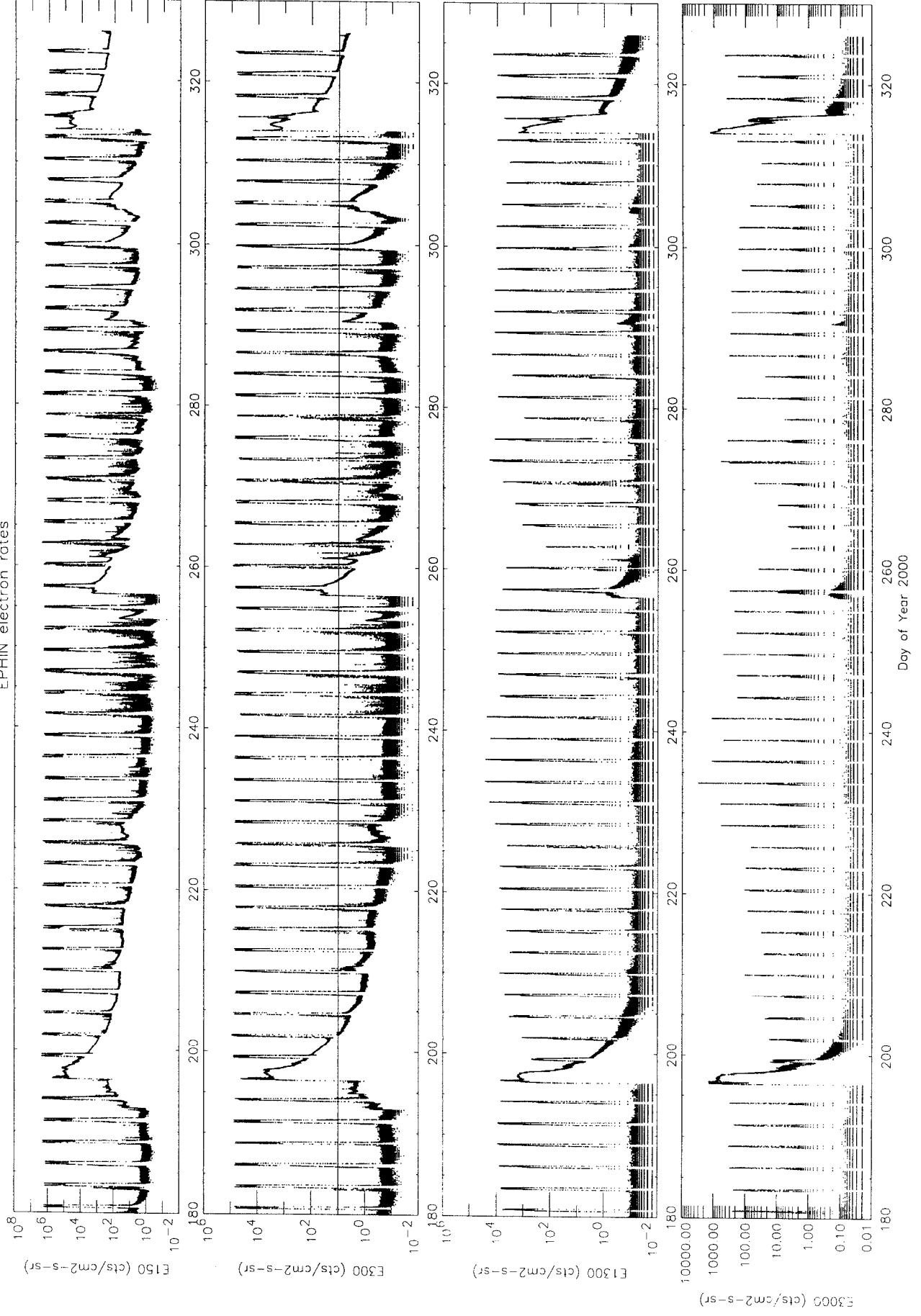
SI Safing Enhancements

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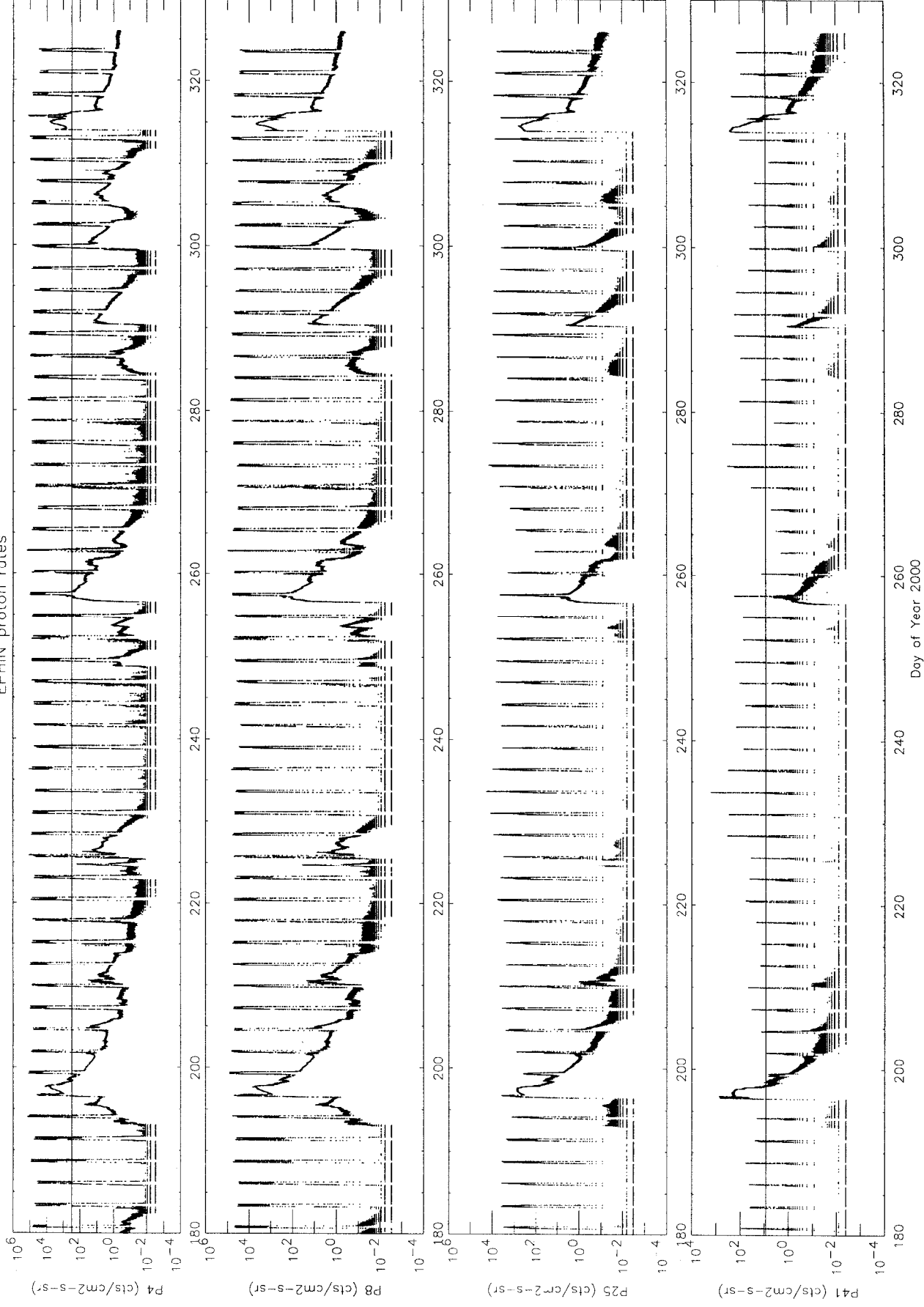
EPHIN Channels and RADMON Trigger Thresholds

<u>Type</u>	<u>Name</u>	<u>Energy Range</u> (MeV)	<u>RADMON Threshold</u> (cts/cm ² -s-sr)
Electron	E150	0.25 – 0.70	
	E300	0.67 – 3.00	
	E1300	2.64 – 6.18	10.0
	E3000	4.80 – 10.4	
Proton	P4	5.0 – 8.3	300.0
	P8	8.3 – 25.0	
	P25	25.0 – 41.0	
	P41	41.0 – 53.0	8.47
Helium	H4	5.0 – 8.3	
	H8	8.3 – 25.0	
	H25	25.0 – 41.0	
	H41	41.0 – 53.0	
Integral	INT	e>8.7, p>53, He>53	

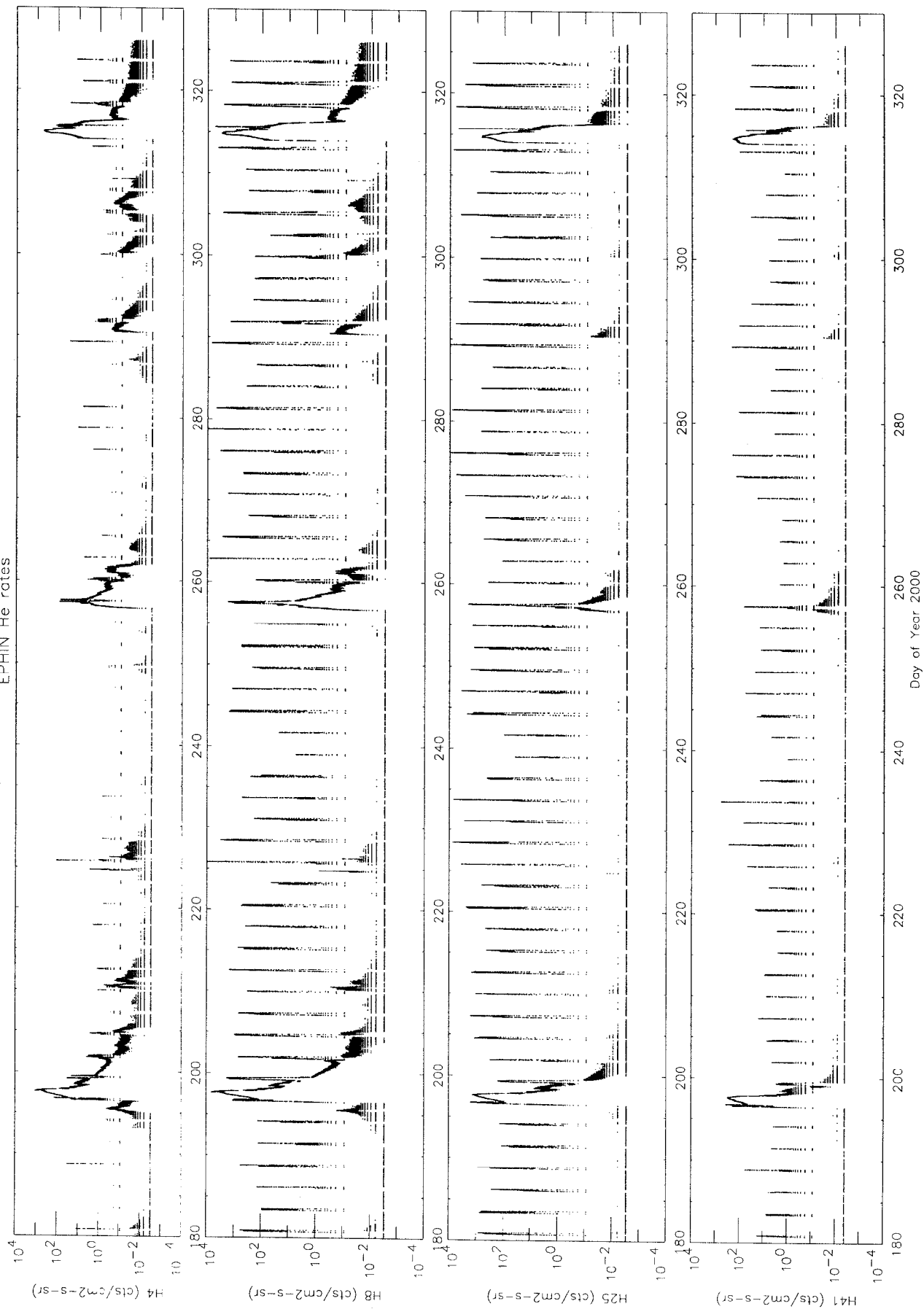
EPHIN electron rates



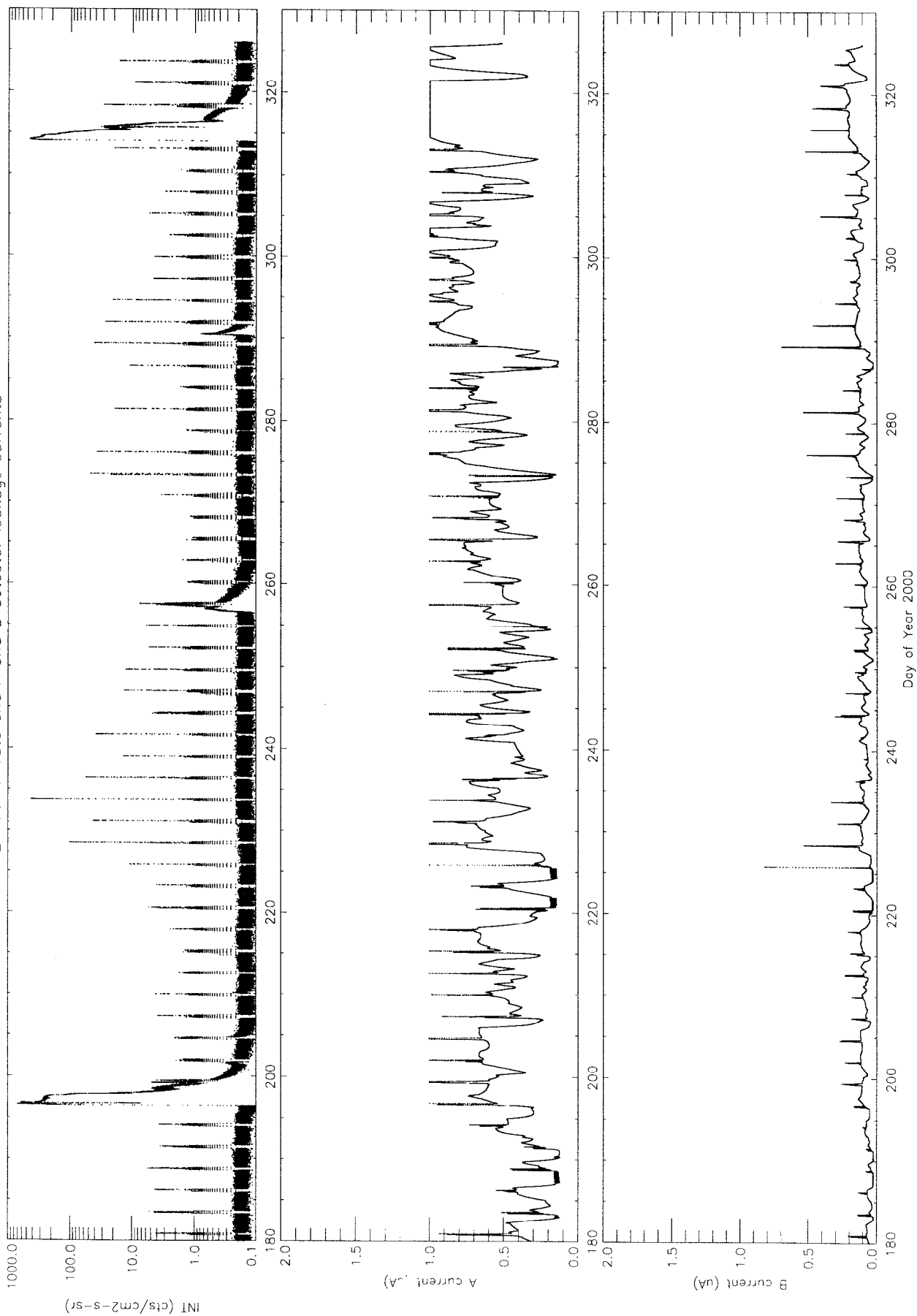
EPHIN proton rates



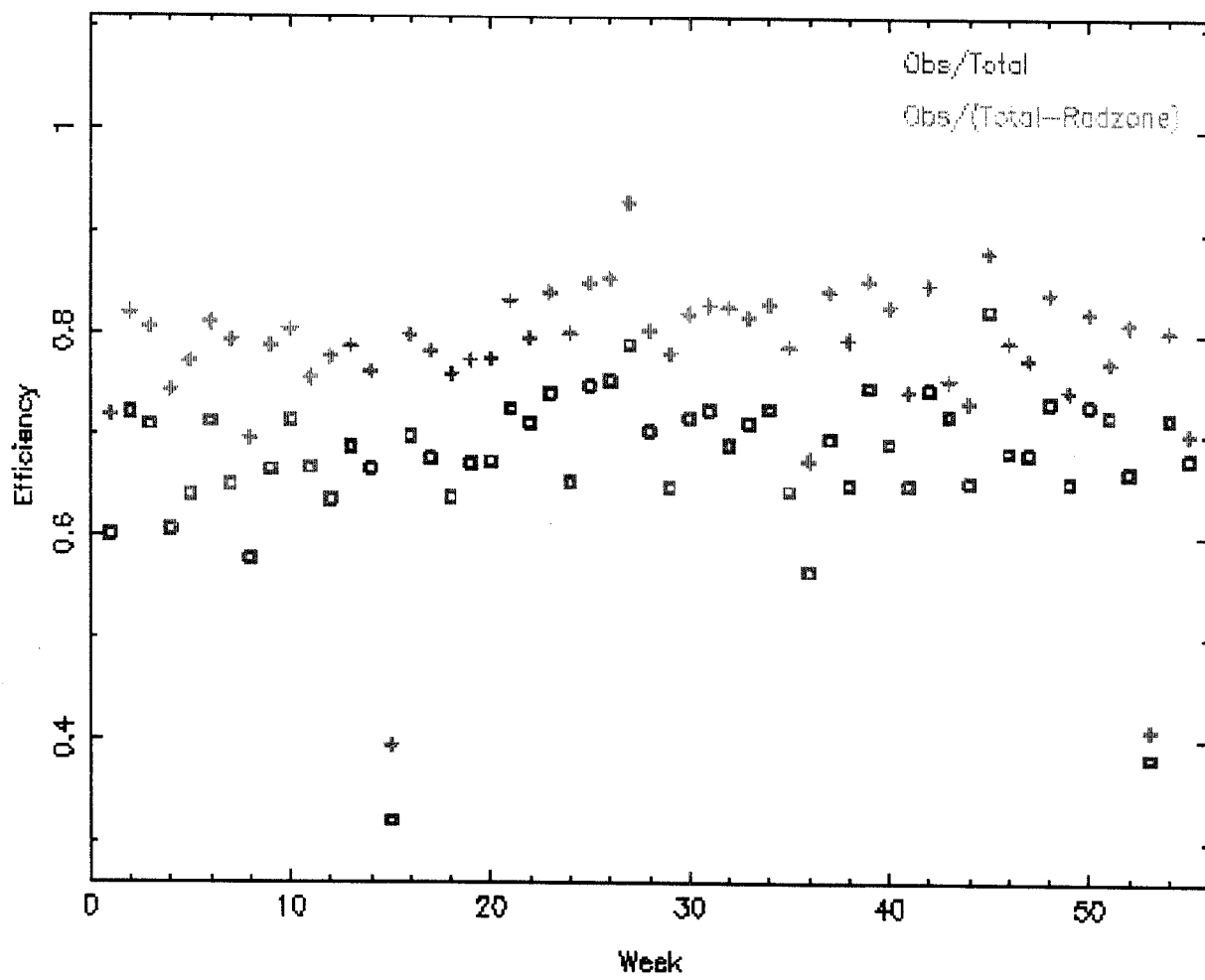
EPHIN He rates



EPHIN INT rate and A and B detector leakage currents



Observing Efficiencies





The Chandra Science Operations Team maintains a series of proton fluence monitors.

Source data:

- Advanced Composition Explorer (ACE): Electron, Proton and Alpha Monitor (EPAM) P3 channel: 112 – 187 keV protons.
- Chandra Radiation Model (CRM)
- Air Force and Costello Kp

Monitors:

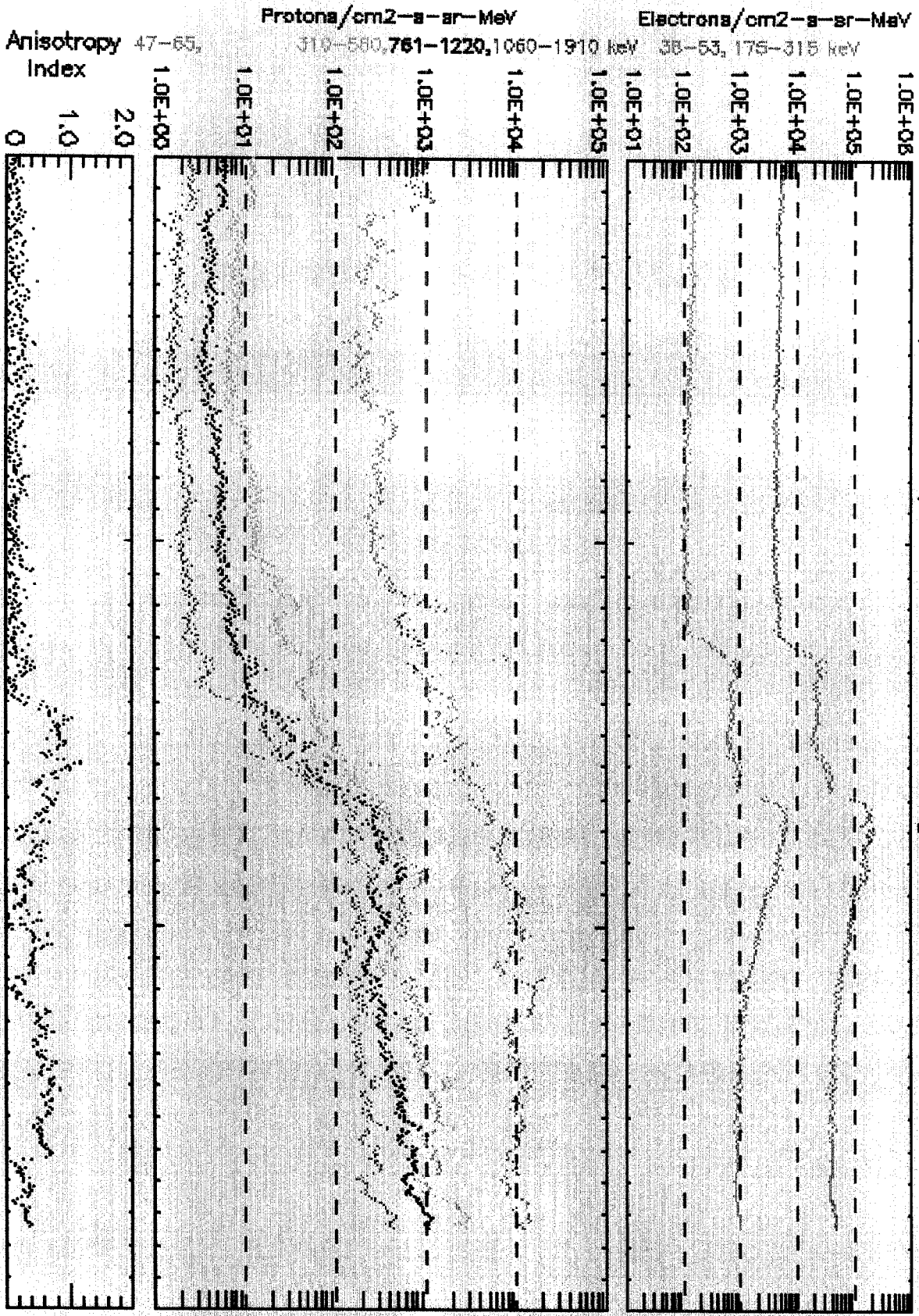
- 2 hour ACE P3 average flux. Alert threshold = $5E4$ cts/cm²-s-sr.
- Chandra orbital fluence. Alert thresholds = $1E9$, $3E9$, $9E9$... cts/cm²-s-sr. ACE EPAM P3 and CRM fluxes are integrated for the duration of each Chandra orbit. ACIS flux estimate is attenuated by SIM and OTG transmission factors (TF):
 - SIM translation table. ACIS: TF = 1, HRC: TF = 0.
 - LETG inserted: TF = 0.5
 - HETG inserted: TF = 0.2
- Air Force and Costello Kp estimates. Variable threshold matrix.

Current studies: correlation of EPHIN P4 with GOES P2 (protons: 4 – 9 MeV).

UTC 2000:33:00:19:35:02 (MJD5555) - 1999-09-13 01:36:00 +00:00 1999-09-13 01:36:00	Current Data:		UTC 2000:33:00:19:35:02 (MJD5555) - 1999-09-13 01:36:00 +00:00 1999-09-13 01:36:00
<p> Chandra Science Directions WWW Computers News Boston Miscellany </p> <p> Bookmark Location: http://cxc.harvard.edu/realtime/cgi-bin/snaps.cgi </p> <p> Back Forward Reload Home Search Netcape Print Security Stop </p> <p> Help </p>	<p> Chandra Science Directions WWW Computers News Boston Miscellany </p> <p> Bookmark Location: http://cxc.harvard.edu/realtime/cgi-bin/snaps.cgi </p> <p> Back Forward Reload Home Search Netcape Print Security Stop </p> <p> Help </p>	<p> Chandra Science Directions WWW Computers News Boston Miscellany </p> <p> Bookmark Location: http://cxc.harvard.edu/realtime/cgi-bin/snaps.cgi </p> <p> Back Forward Reload Home Search Netcape Print Security Stop </p> <p> Help </p>	<p> Chandra Science Directions WWW Computers News Boston Miscellany </p> <p> Bookmark Location: http://cxc.harvard.edu/realtime/cgi-bin/snaps.cgi </p> <p> Back Forward Reload Home Search Netcape Print Security Stop </p> <p> Help </p>

ACE RTSW (Estimated) EPAM

Begin: 2000-11-23 00:00:00UTC



start DOY: 328

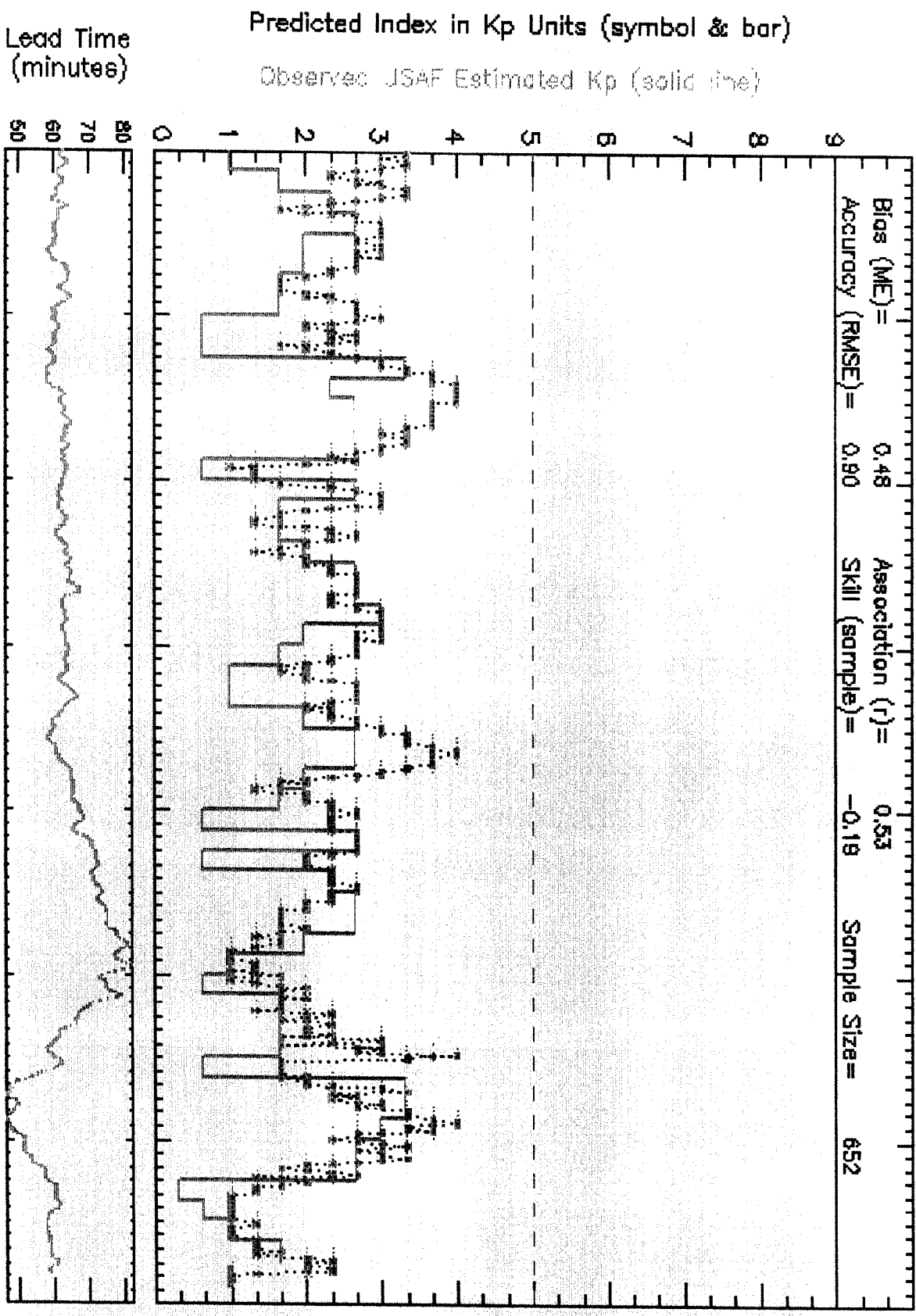
UTC(days)

created: 2000-11-25 18:04:10

Costello Predicted Activity Index

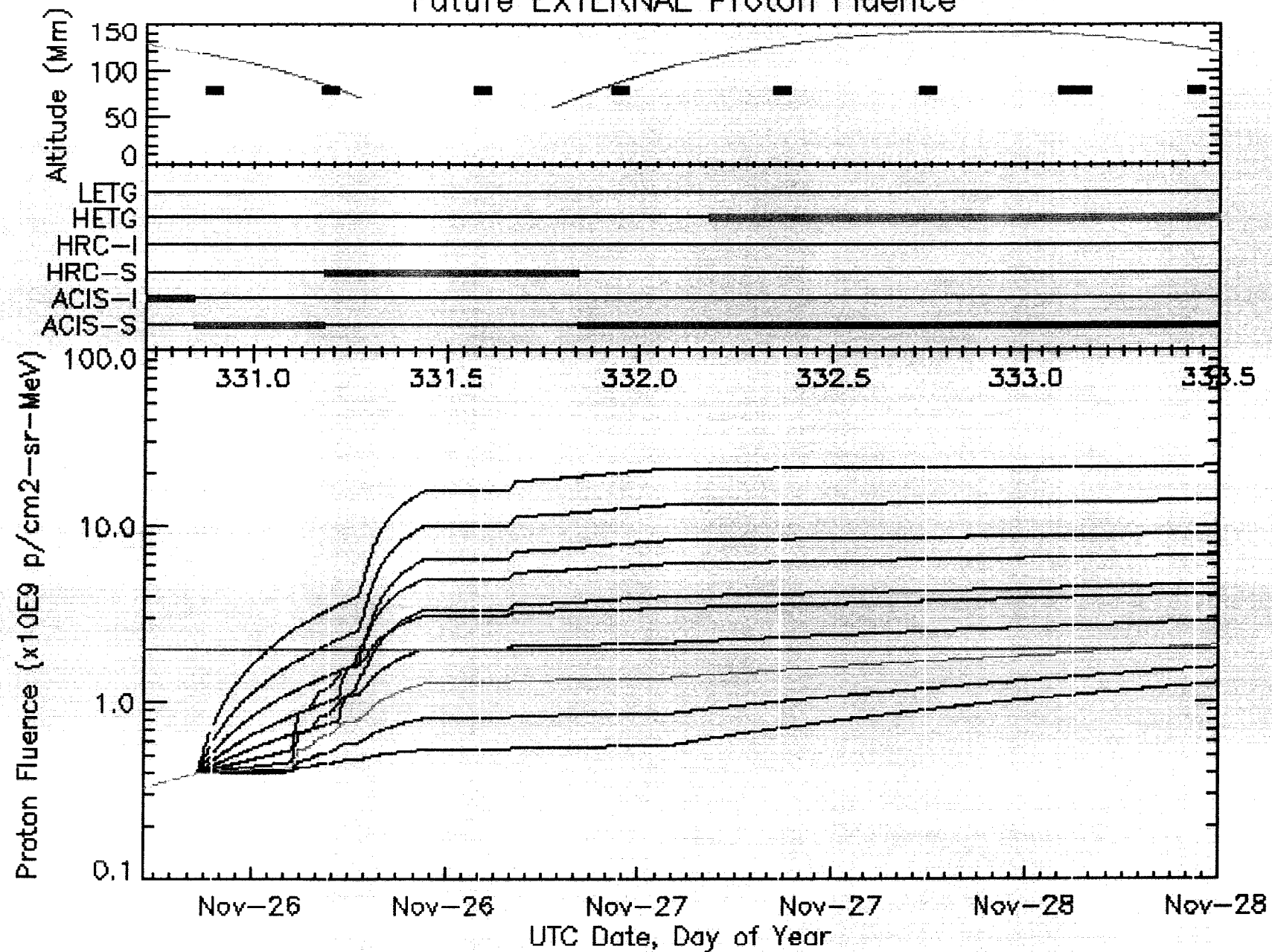
Begin: 2000-11-19 00:00:00UTC

Bias (ME) = 0.48 Association (r) = 0.53
Accuracy (RMSE) = 0.90 Skill (sample) = -0.19 Sample Size = 652

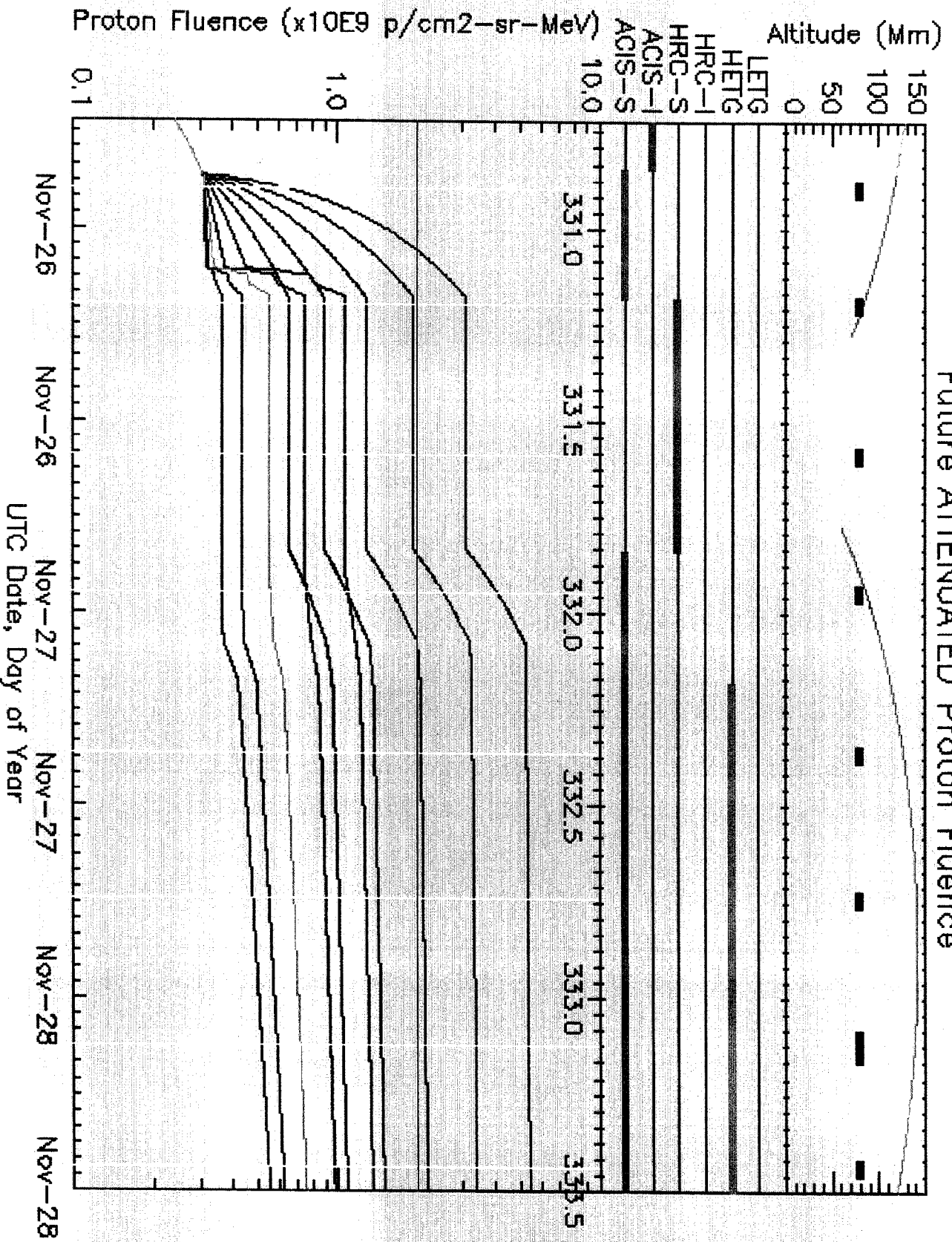


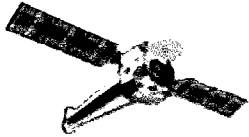
start DOY: 324 status = 0.123 UTC(days) created: 2000-11-25 19:36:07UTC

Future EXTERNAL Proton Fluence



Future ATTENUATED Proton Fluence





Chandra Radiation Model

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Chandra Project Science is leading a team to develop and implement an improved model for low energy proton fluxes in the Chandra orbit: the Chandra Radiation Model (CRM). Model development is led by Bill Blackwell (Sverdrup Technology Inc.) and the Chandra Radiation Working Group.

Objectives:

- Improve Chandra science observing efficiency, with low risk of RADMON triggers from trapped radiation.
- Improve real time predictions of Chandra proton fluence, for health and safety decision making.
- Refine understanding of the effects of Chandra particle environment on ACIS, for improved budgeting and control of lifetime accumulated damage.

Current Chandra mission planning and scheduling uses the NASA AP-8/AE-8 radiation models. AP-8 provides poor fidelity above geosynchronous orbit and poor modeling of 100keV protons. AE-8 is used for scheduling science operations in each Chandra orbit. Mismatches between AE-8 predictions and observed EPHIN rates have resulted in required 10ksec “padding” on both the predicted radiation entry and exit times.

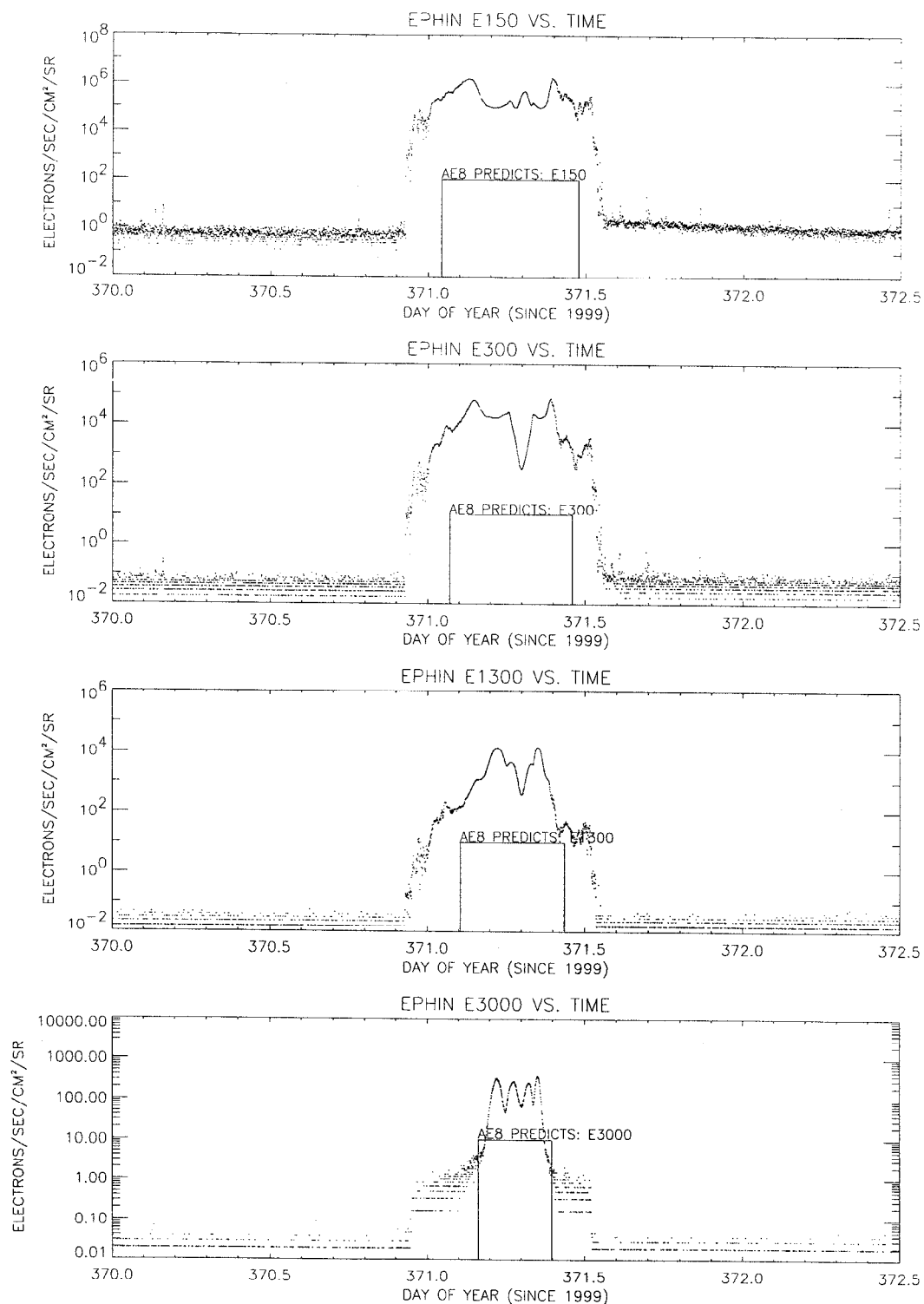
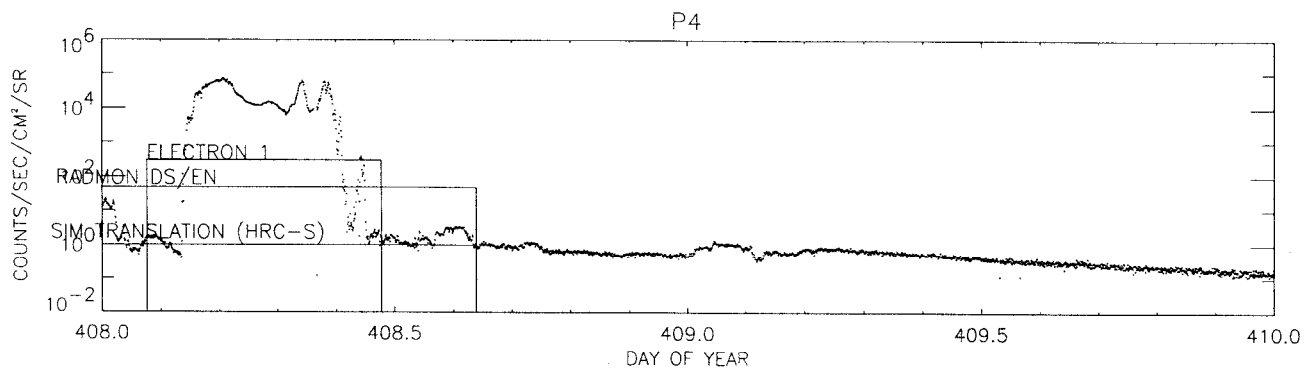
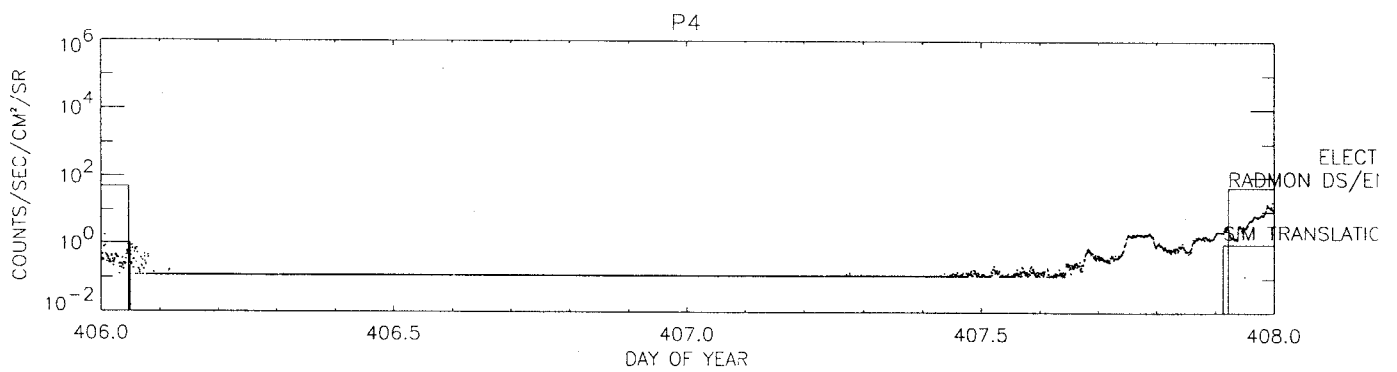
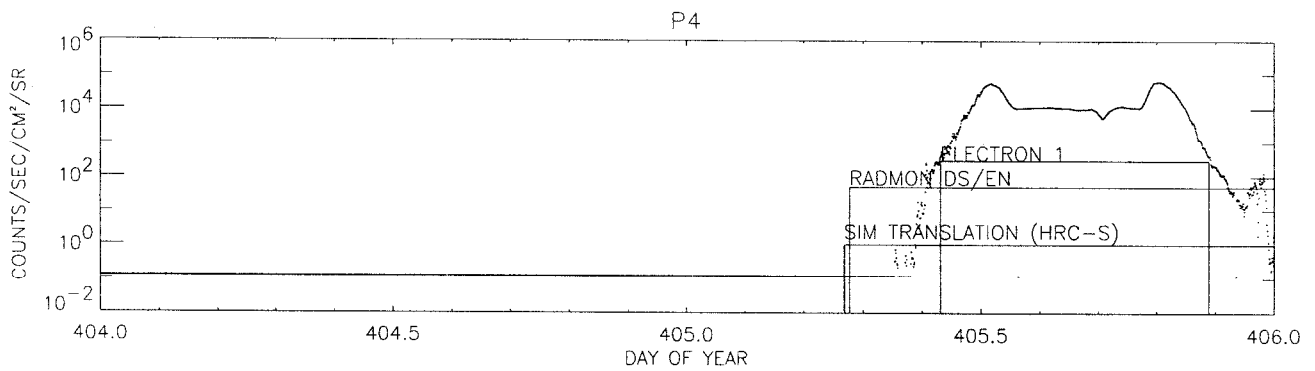
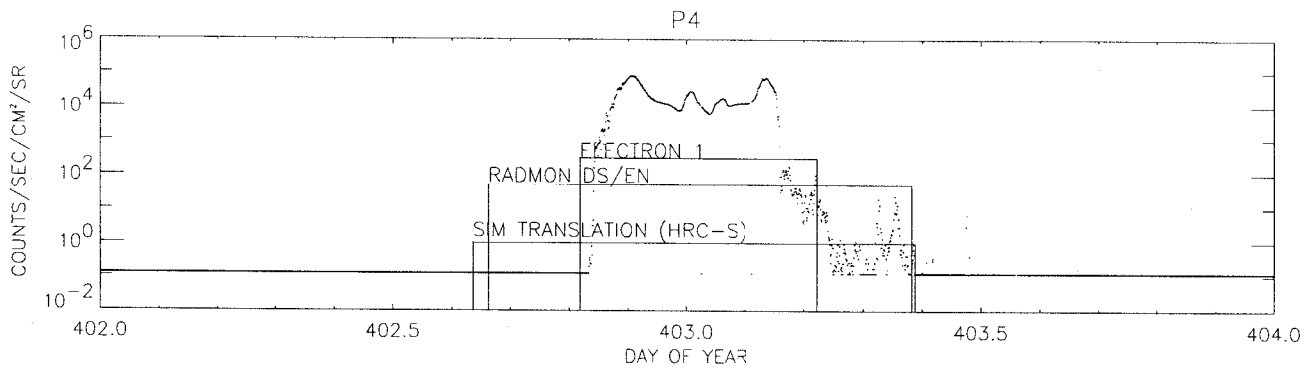
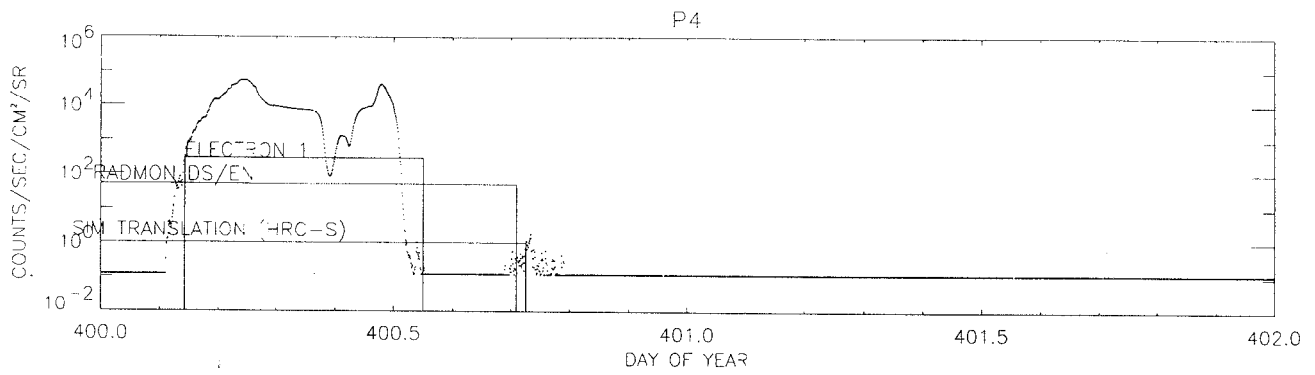


Figure 5. AE-8 radiation belt model predictions against EPHIN data on Day 371. Note AE-8 predictions are based on an integral above an energy threshold (specified within the text) whereas the EPHIN electron channel has a much narrower energy window.





Chandra Radiation Model

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The CRM is a data-drive model. CRM characteristics:

- Model for 100 – 200 keV protons, compatible with ACE EPAM P3 data.
- 9 – 30 Re geocentric altitude coverage.
- Indexed by Geocentric Solar Magnetospheric (GSM) coordinates and Kp.
- Flux predictions identified by region: magnetosphere, magnetosheath, solar wind for improved estimation of Chandra global radiation environment.
- Probabilistic flux estimates: mean, 50%, 95% estimates.
- Multiple species and energies: p: 100 – 200 keV, p > 1 MeV, He, CNO.

CRM data source: 5 year dataset (1995 – 2000) from the Ion Composition Subsystem (ICS) of the Energetic Particles and Ion Composition (EPIC) instrument on the Japan ISIS/NASA Geotail satellite. Ion and electron spectrograms from the Comprehensive Plasma Instrument (CPI) Hot Plasma Analyzer (HPA) provided the region identification.

All Regions

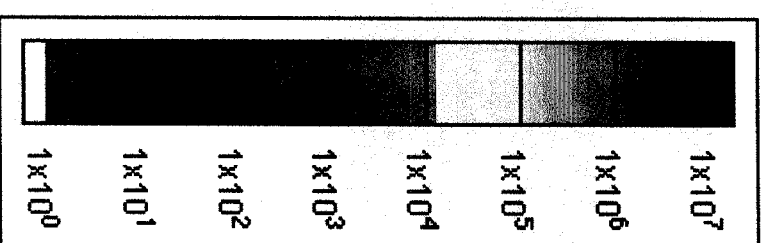
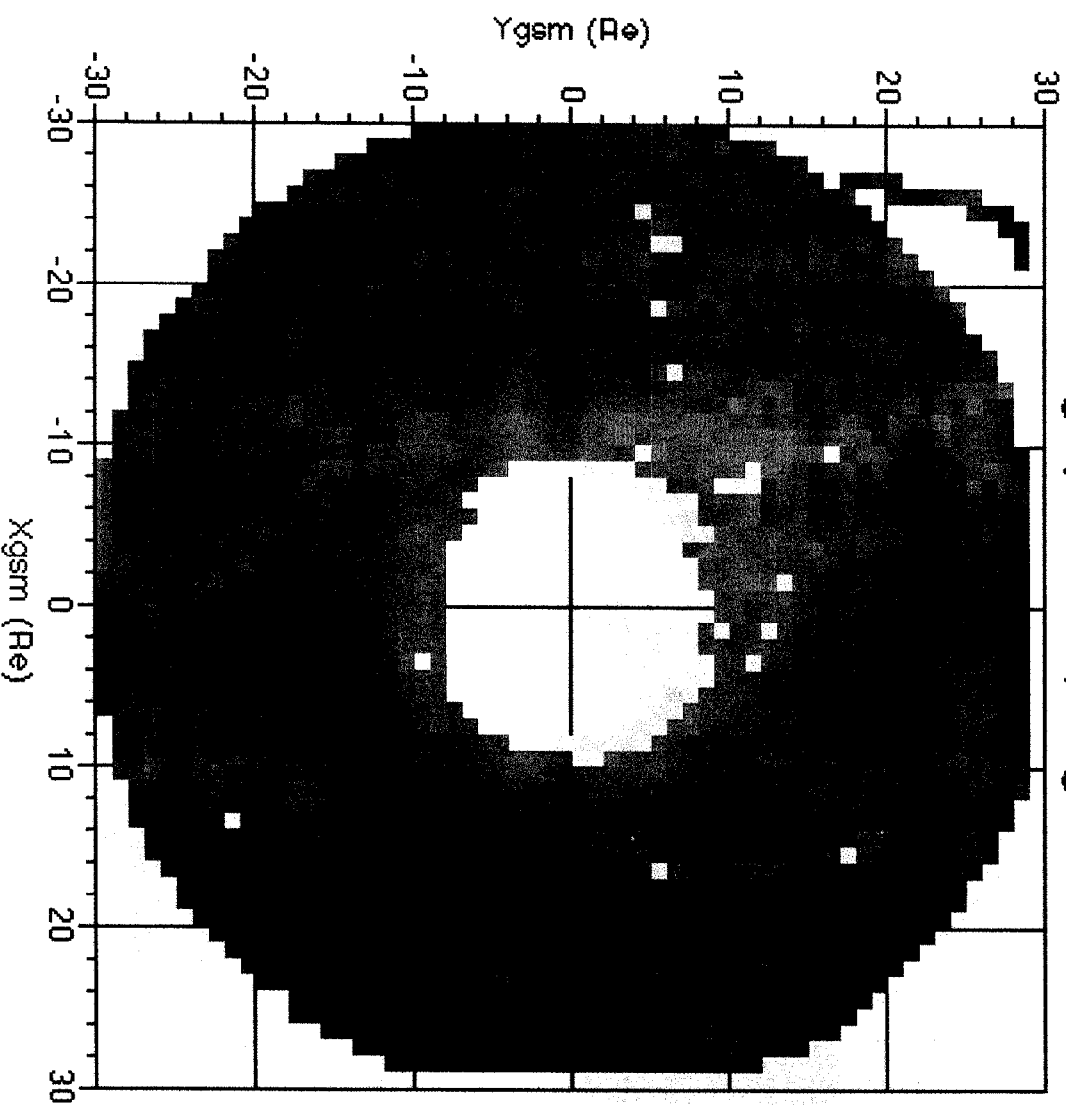
XY Flux Slice [protons/(cm²-sec-sr-MeV)]

Average of all Z-values; 100 - 200 keV protons

(Kp 0-2; Includes Solar Event Particles)

01-01-1995 to 04-30-2000

Avg Kp = 0.915; Avg Dst = -6.05



All Regions

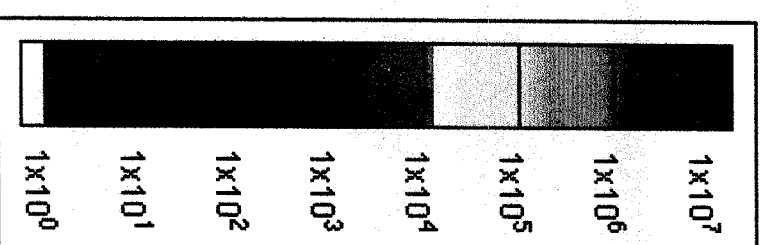
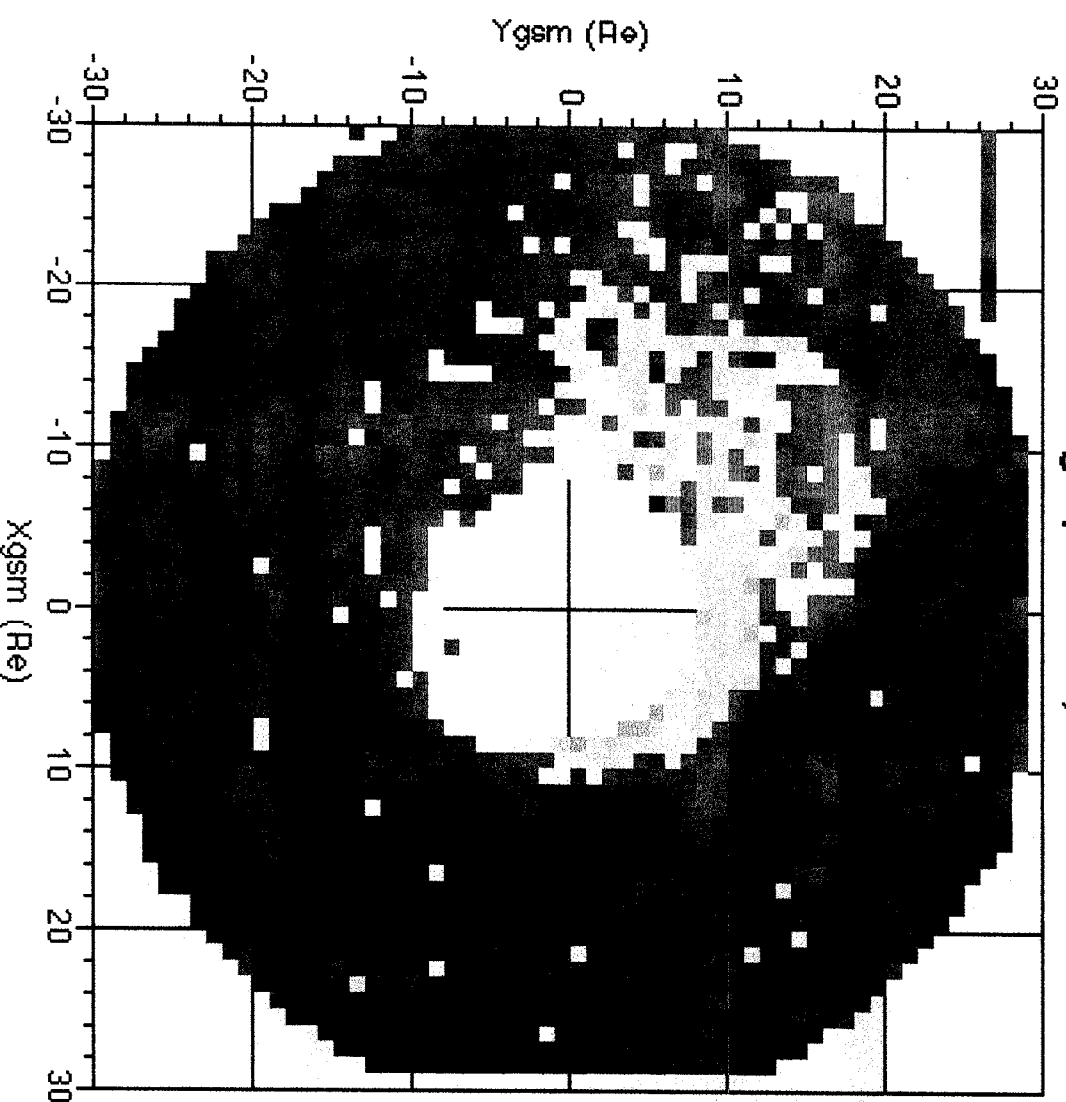
XY Flux Slice [protons/(cm²-sec-sr-MeV)]

Average of all Z-values; 100 - 200 keV protons

(Kp 2-4; Includes Solar Event Particles)

01-01-1995 to 04-30-2000

Avg Kp = 2.63; Dst = -17.33



All Regions

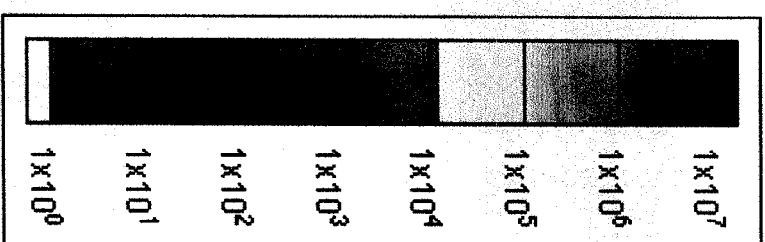
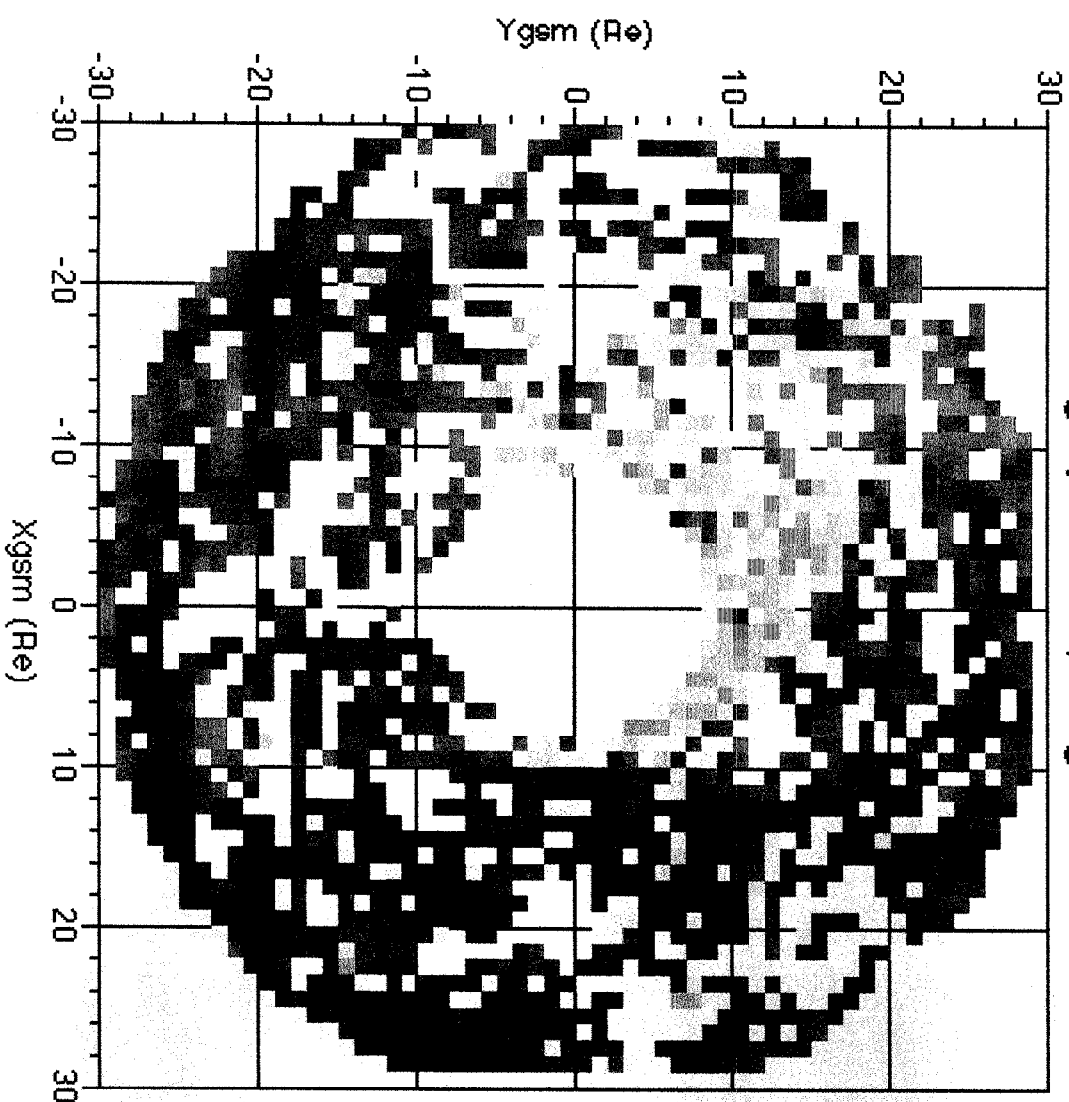
XY Flux Slice [protons/(cm²-sec-sr-MeV)]

Average of all Z-values; 100 - 200 keV protons

(Kp 4-6; Includes Solar Event Particles)

01-01-1995 to 04-30-2000

Avg Kp = 4.46; Avg Dst = -35.82



All Regions

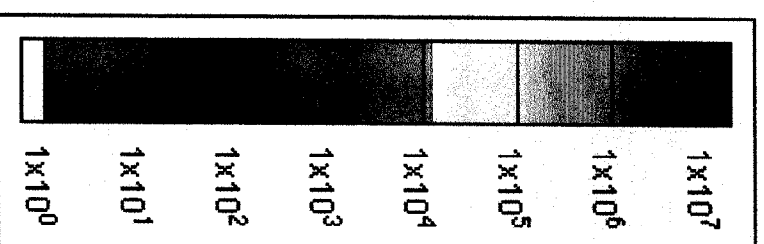
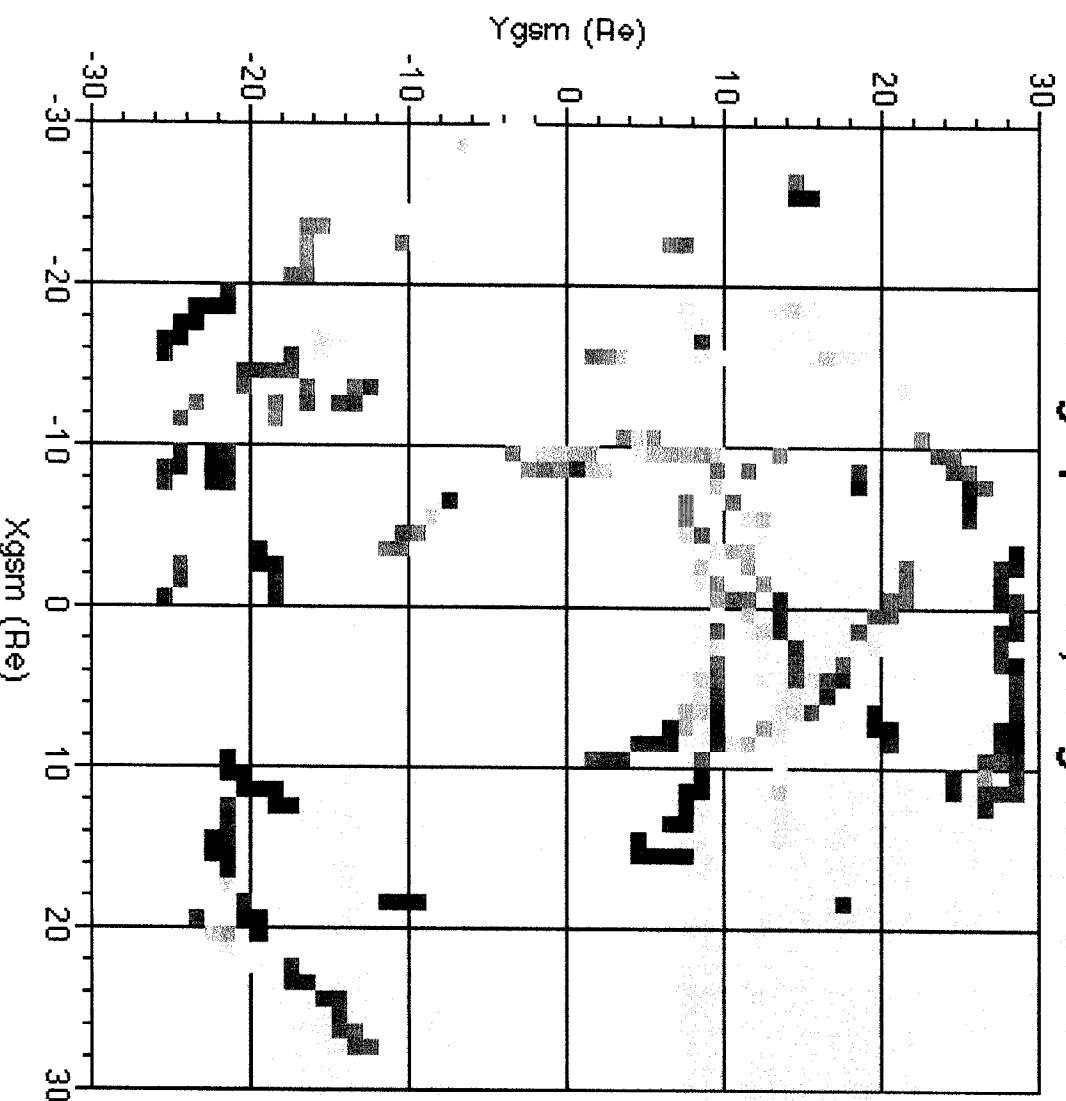
XY Flux Slice [protons/(cm²-sec-sr-MeV)]

Average of all Z-values; 100 - 200 keV protons

(Kp 6-9; Includes Solar Event Particles)

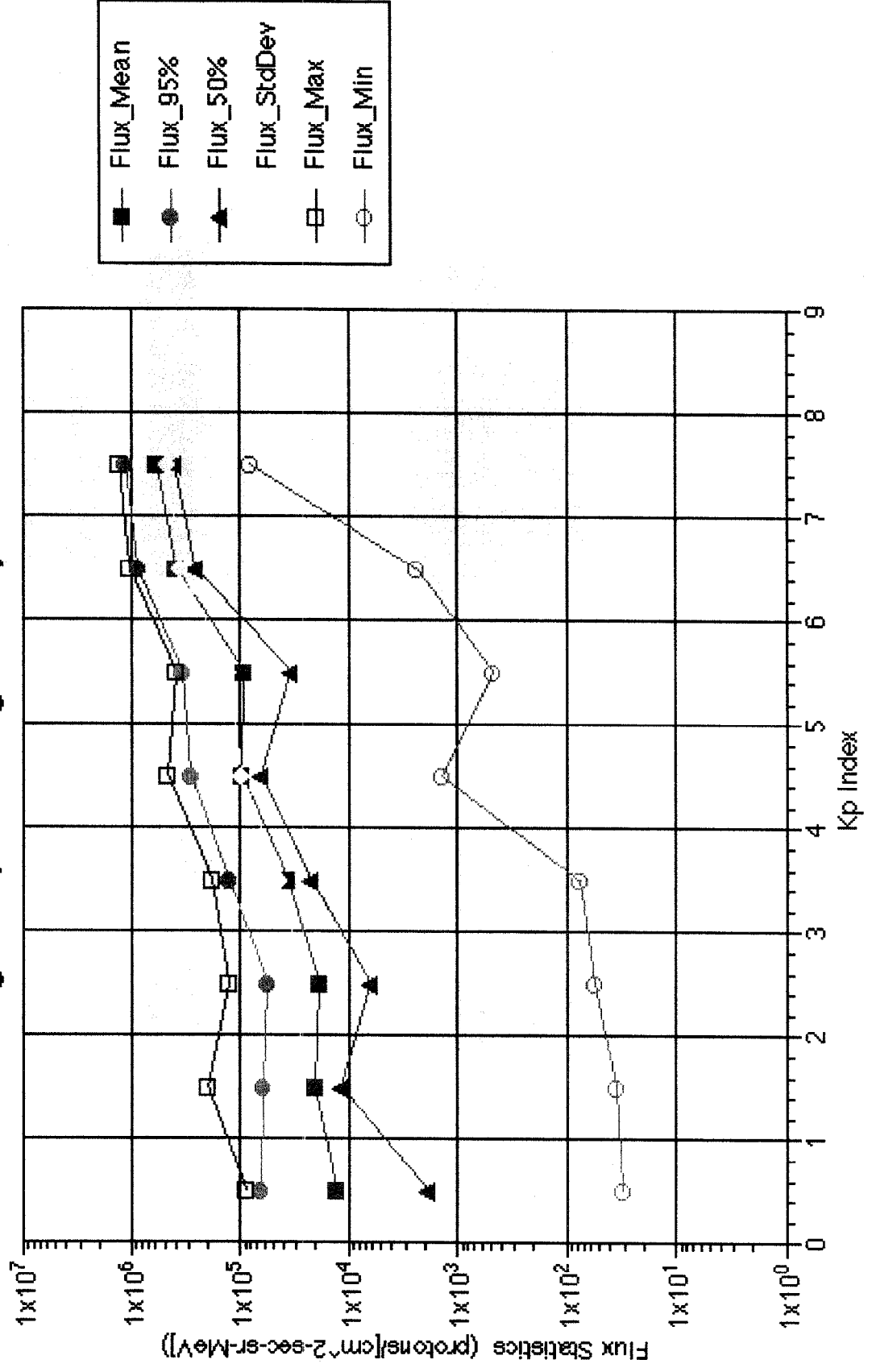
01-01-1995 to 04-30-2000

Avg Kp = 6.63; Avg Dst = -89.19



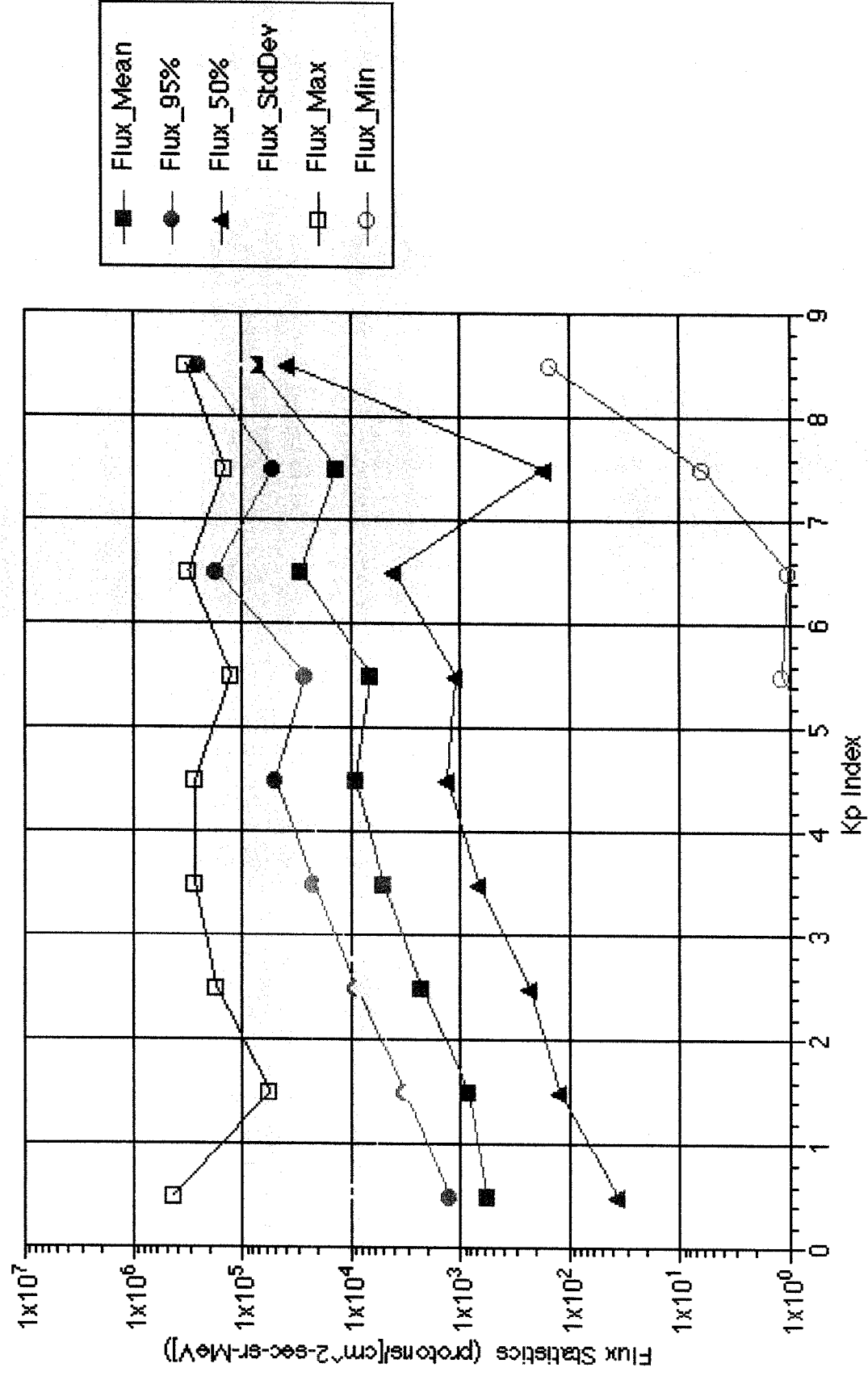
Sector 1: Database (-10<X<-6; -6<Y<+4; All Z)

Magnetosphere Region Only



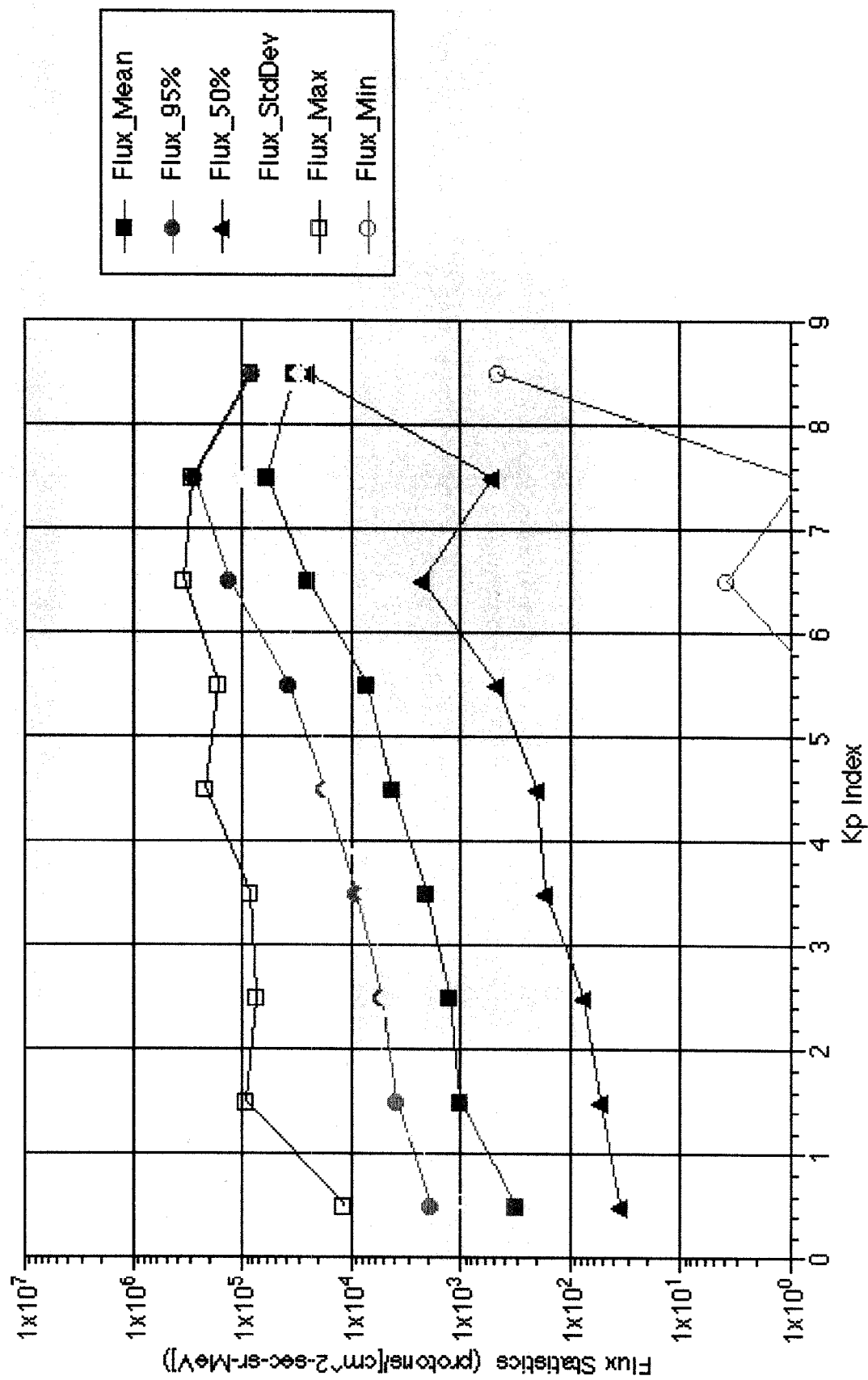
No Sectoring; All Z: Database

Magnetosheath Region Only

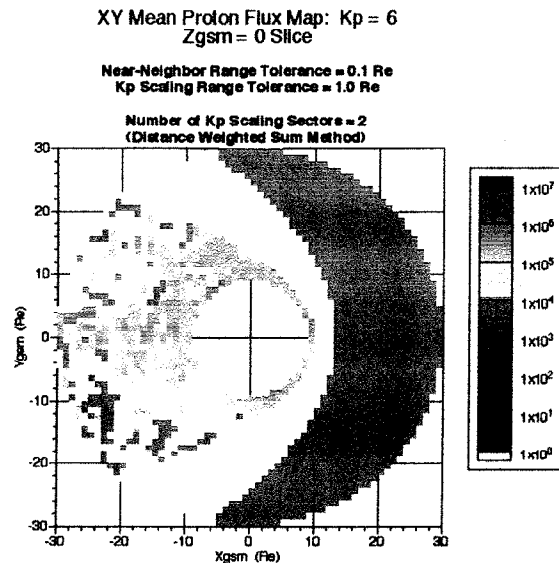


No Sectoring; All Z: Database

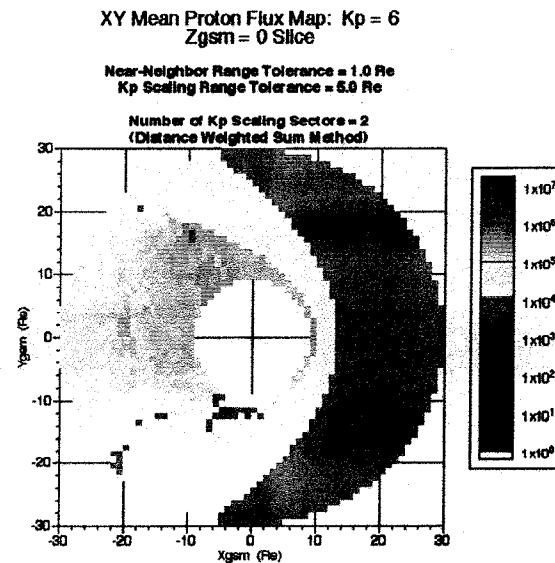
Solar Wind Region Only



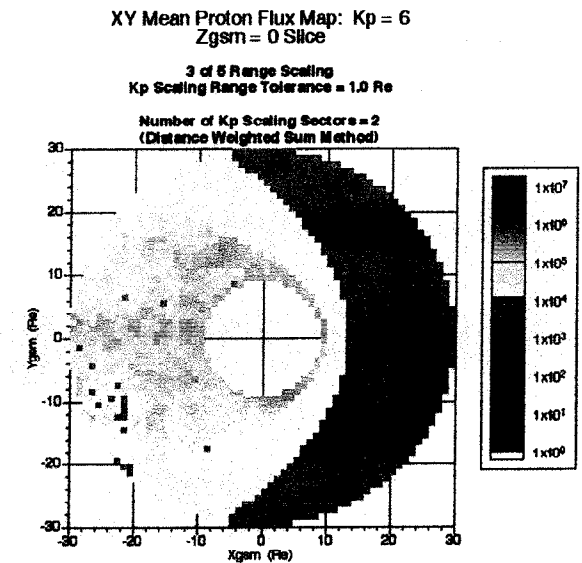
Preliminary CRMFLX Smoothing Results



No Data Smoothing



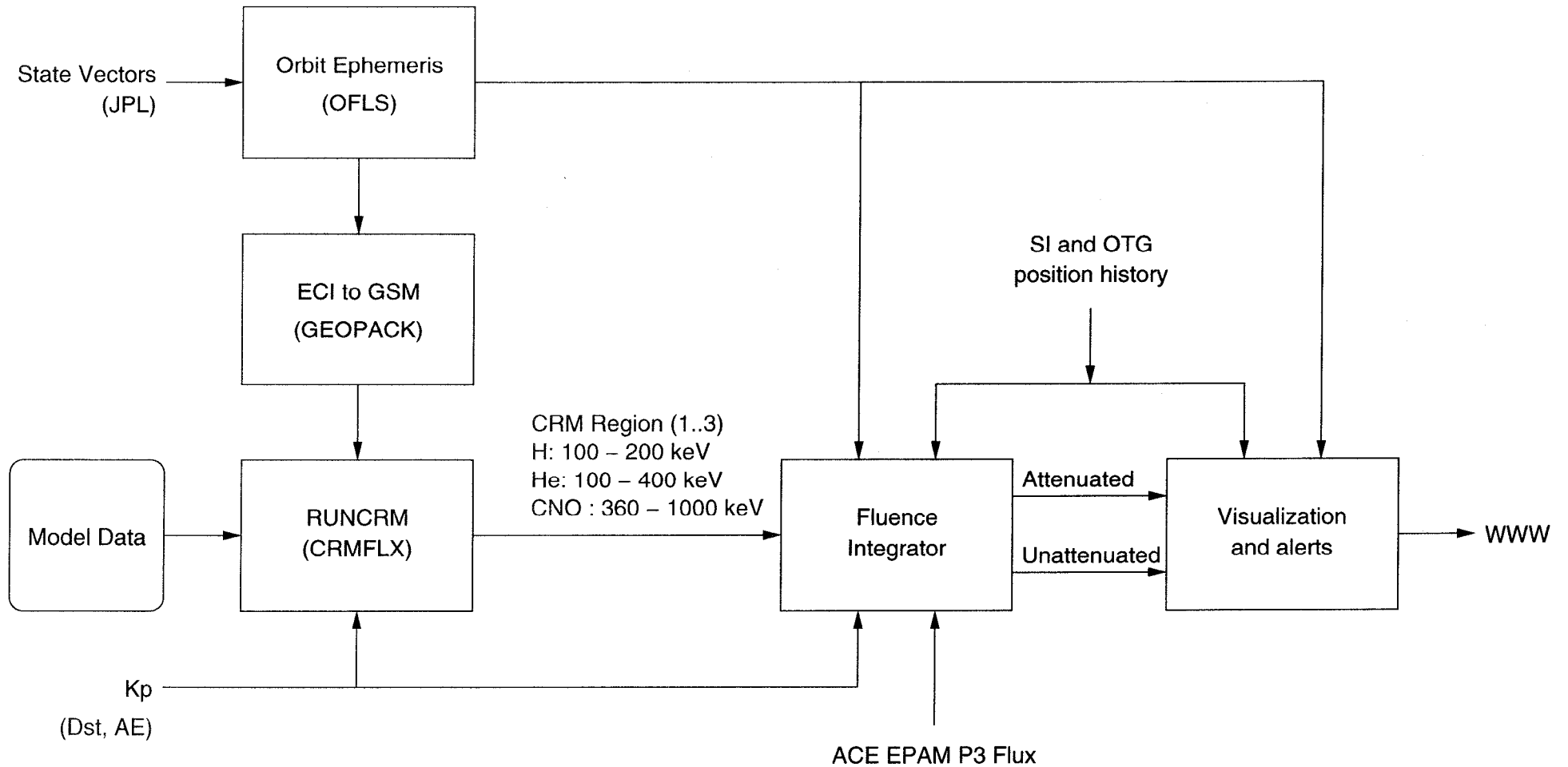
Spatial Averaging

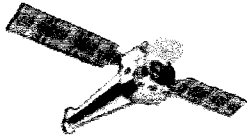


3-of-5 Spike Rejection

- A number of different smoothing algorithms are now being tested in CRMFLX
 - Variations of spatial averaging and spike rejection
 - Care must be taken so to not throw away physical variations
- Much of the data scatter will be removed by performing better data correlation
 - Include solar wind conditions as well as geomagnetic activity (Kp index)
 - Data correlation is next priority task

Chandra Radiation Model (CRM) realtime and predictive system





Chandra Radiation Model

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CRM development status:

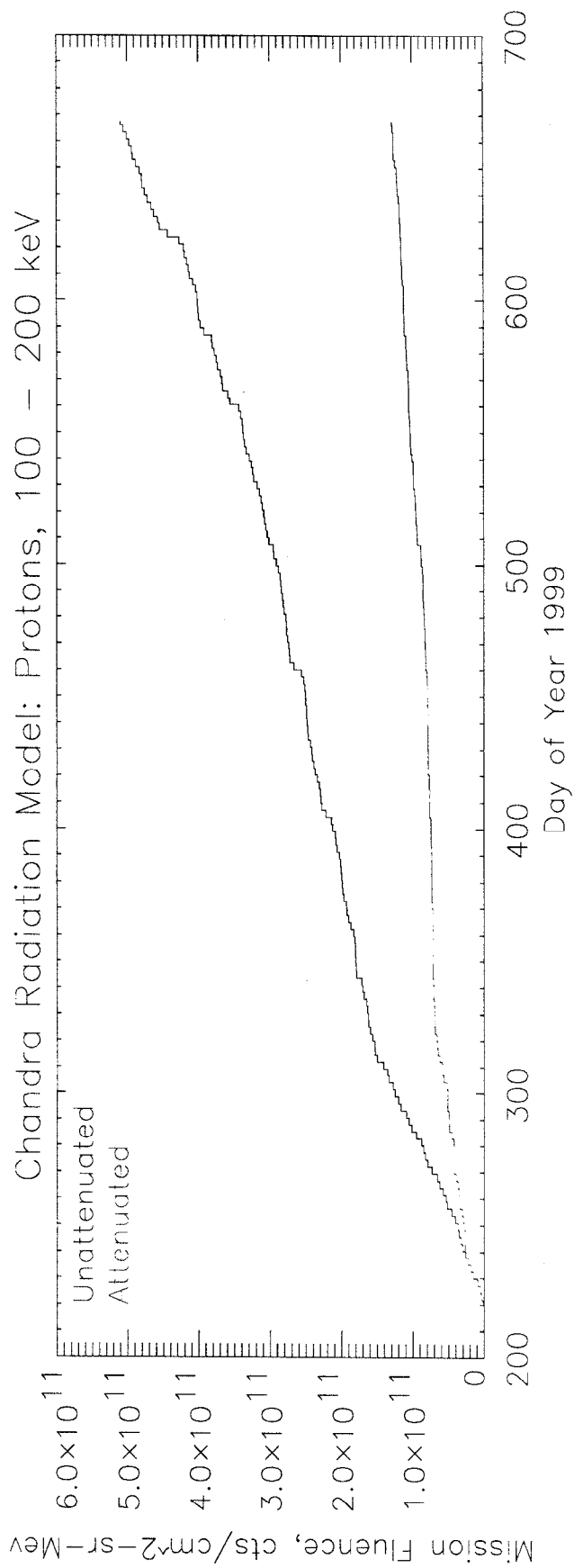
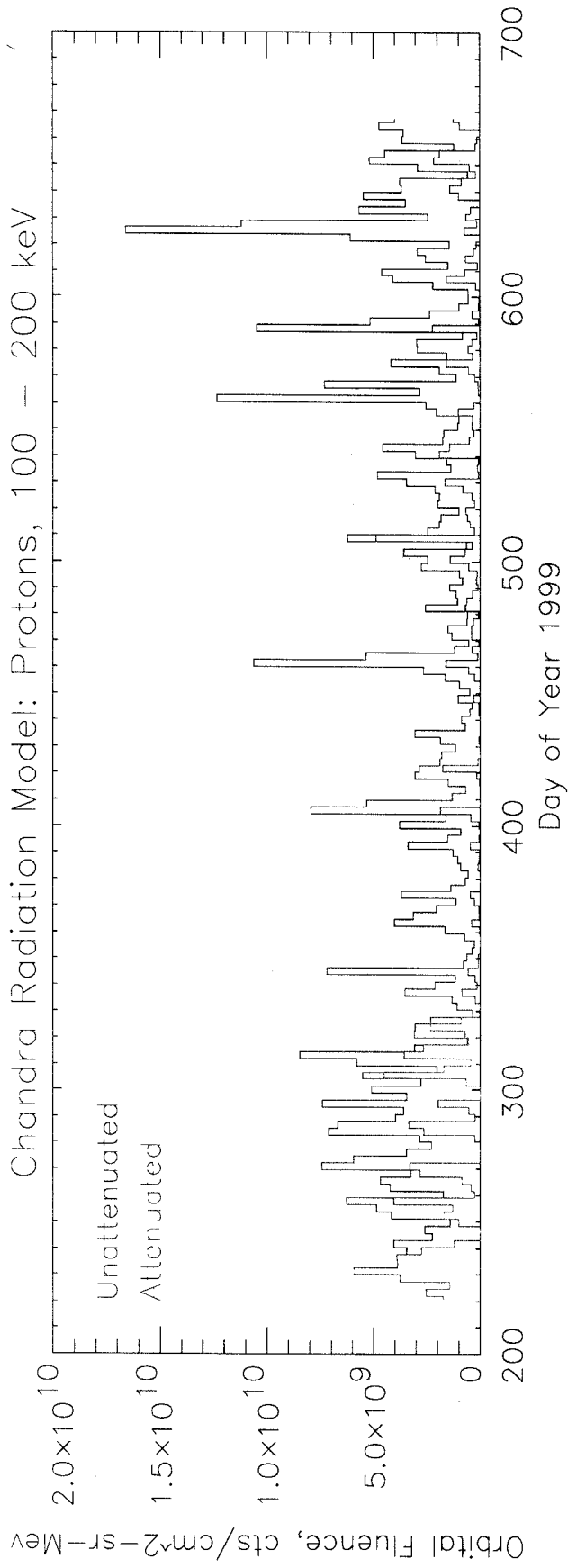
- Version 1.0 released 8 September 2000.
- Version 1.1 released 16 October 2000. Data interpolation errors fixed. 50% and 95% flux estimates corrected.

Usage:

- Version 1.1 implemented by the Science Operations Team for real time integration of Chandra flux and orbital fluence and prediction of future fluence.
- Development in progress of orbital start science and stop science scheduler for Chandra Operations Control Center Off-Line System (OFLS). Objective: optimize science time per orbit while maintaining fluence within specified limits.

Issues:

- Data model empty below 9 Re: incomplete measure of Chandra radiation history.
- Data interpolation; sparse data at high Kp.
- $P > 1$ MeV proton data are needed to allow OFLS testing based on comparison with existing AE-8 and AP-8 models.
- Devolution of OFLS science time scheduling to CRM.





Data Sources

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NASA Space Environment Center:

<http://www.sec.noaa.gov/index.html>

ACE:

Data: **http://sec.noaa.gov/ace/ACERTsw_data.html**

ACE Science Center: **<http://www.srl.caltech.edu/ACE/ASC>**

Geotail:

<http://www-spof.gsfc.nasa.gov/istp/geotail>

Chandra Project Science:

<http://xanth.msfc.nasa.gov/xray/axafps.html>

Chandra Radiation: **<http://wwwastro.msfc.nasa.gov/xray/ACIS/fluence>**

Chandra Science Operations Team:

<http://asc.harvard.edu/mta/sot.html>